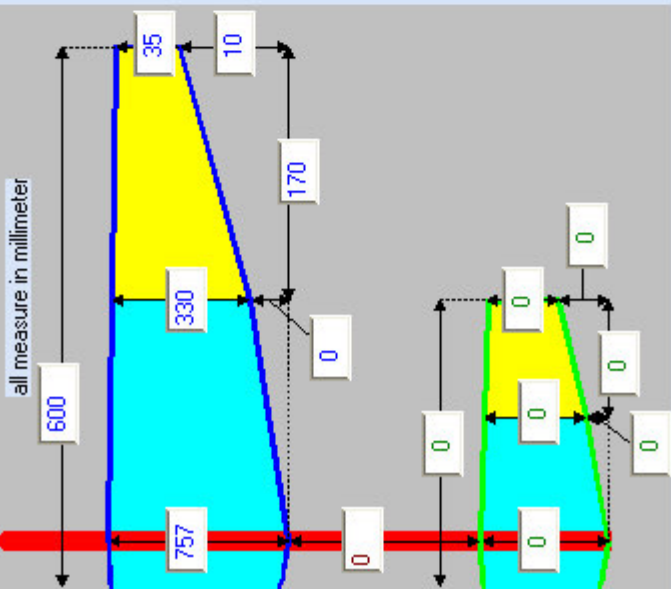


C:\Documents and Settings\Dennis L Payne\My Documents\My Model information\Aerodynamics\area by dm - 2. wing: total: tailplane: aerodynamic programs\CG calculations\Schwer\Bae Concorde.dat



mean aerodyn. chord:	464.9	tailplane:	0.0	total:	20.3
aspect ratio:	1.8	wing:	20.3		
lift slope:	2.4	area by dm - 2:	464.9		
aerodynamic center:	407.2	lever arm:	0.000		
aerodynamic center by %:		down wash factor:	0.000		
		centre of gravity by stab.:	8 %		370.0
		centre of gravity by stab.:	4 %		388.6

